

DRIMI - DOTTORATO DI RICERCA IN INGEGNERIA MECCANICA ED INDUSTRIALE



TURBOPUMP IMPELLER DESIGN BY OPEN-SOURCE CFD AND SHAPE OPTIMIZATION SOFTWARE

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2nd year presentation - Brescia, 18/10/2016

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Introduction

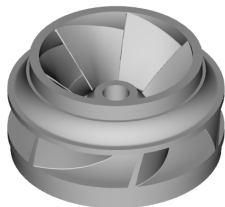
- Turbopumps are very used in industry
- Design techniques to achieve the optimal geometry for the given working point are too expensive
- The efficiency and the pressure ratio of a centrifugal pump greatly depend on the shape of the impeller and the flow passage between the hub and shroud
- Other components of the pump play a role in the performance definition, however only the impeller is considered in this study

Introduction

- An existing and well known impeller geometry is chosen as starting design
- The starting design has been re-expressed in terms of Bezier polynomials
- The efficiency and pressure ratio of the turbopump impeller are evaluated through computational fluid dynamics
- Direct single objective genetic algorithm optimization is compared to a surrogate based optimization technique
- Open-source software are used in order to fully exploit the available hardware without requiring software licenses
- The objective of the optimization methods proposed is to make the turbopump impeller design automatic, robust and optimal

Methodology: impeller shape definition (1/3)

- mixed flow pump
- experiments by Boccazzi *
- geometry through data points for hub, shroud and blade



Original impeller.

Geometry and results of the experiments:

$$D_{out} = 0.224 \text{ m}$$

$$\omega = 55.4 \text{ rad/s}$$

$$\omega_s = 1.08$$

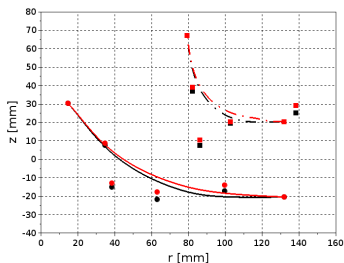
$$\phi = 0.31$$

$$\psi = 0.44$$

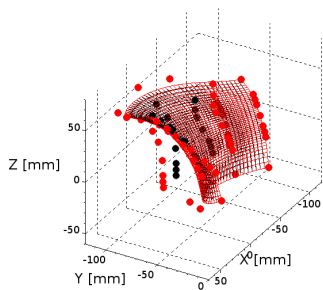
$$\eta = 0.75$$

* Boccazzi, A., Miorini, R., Sala, R., Marinoni, F. (2009), *Unsteady Flow Field in a Radial Pump Vaned Diffuser*, 8th European Conference on Turbo-machinery, Graz, Austria, Mar. 23-27, pp. 1103-1112

Methodology: impeller shape definition (2/3)

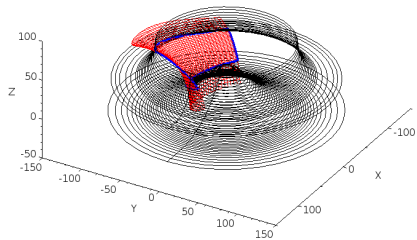


Hub and shroud definition
through Bezier curves

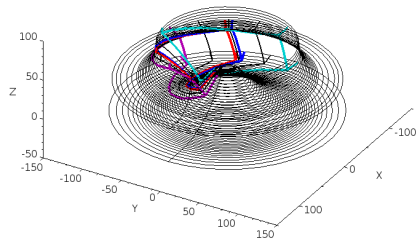


Blade definition through
Bezier surface

Methodology: impeller shape definition (3/3)



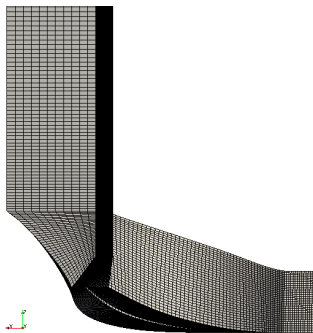
detail of blade intersection
with hub and shroud



detail of fluid domain
preparation

Methodology: computational setup

- fully hexa grid through blockMesh
- incompressible flow
- steady-state condition
- 3D RANS equations
- SST k-omega turbulence model
- $\phi=0.31$ prescribed
- $\psi=0.45$ and $\eta=0.90$ through OpenFOAM



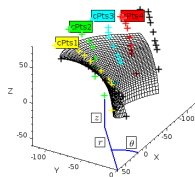
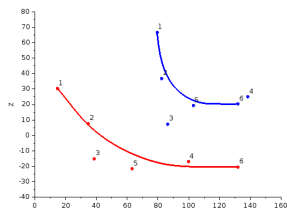
Grid for the CFD calculation

Methodology: optimization setup (1/2)

variables [mm]	zH1	zH2	zH3	zH4
IP	0.0	0.0	0.0	0.0
LB	-5.0	-5.0	-5.0	-5.0
UB	5.0	5.0	5.0	5.0

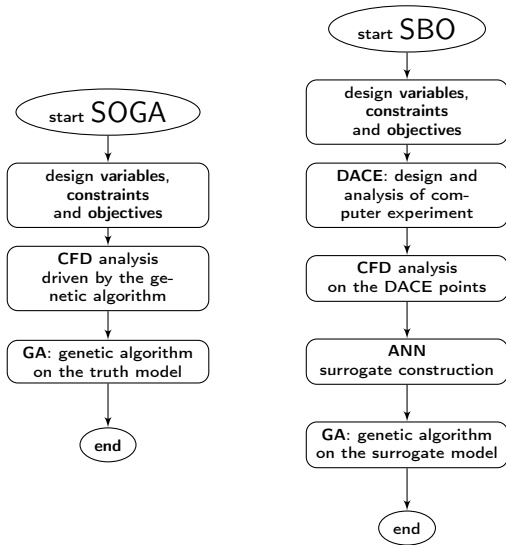
variables [mm]	zS1	zS2	zS3	zS4
IP	0.0	0.0	0.0	0.0
LB	-5.0	-5.0 <td -5.0	-5.0	
UB	5.0	5.0	5.0	5.0

variables [rad]	$\theta B1$	$\theta B2$	$\theta B3$	$\theta B4$
IP	0.0	0.0	0.0	0.0
LB	-0.2	-0.2	-0.2	-0.2
UB	0.5	0.5	0.5	0.5



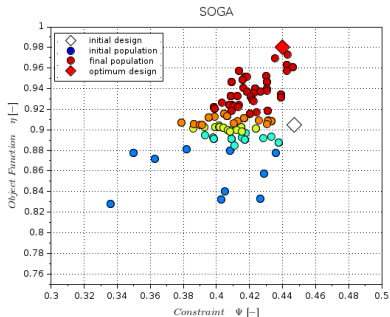
responses	type	sense	LB	UB
η	objective	max	-	-
ψ	NL-IC	-	0.38	0.52

Methodology: optimization setup (2/2)

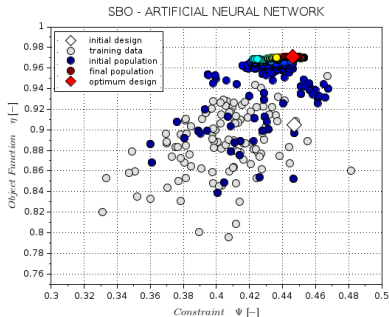


Results

Optimization methods behaviour



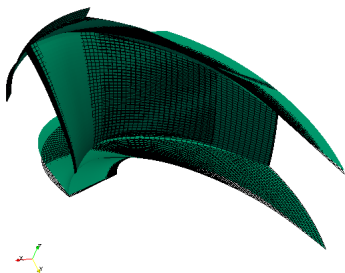
SOGA



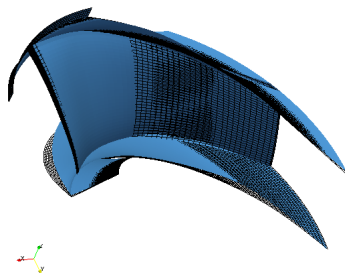
SBO

Results

Detail of blade suction side: comparison between original impeller (black wireframe) and best solution after optimization



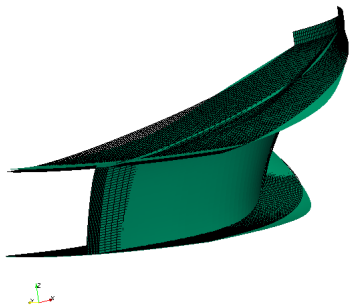
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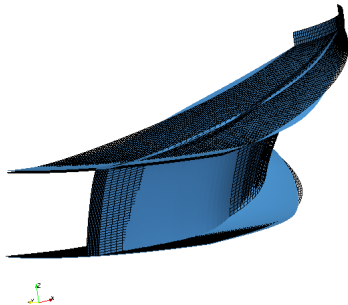
SBO

Results

Detail of blade pressure side: comparison between original impeller (black wireframe) and best solution after optimization



SOGA



SBO

Conclusion and future work

- Two fully automated optimization methods have been presented for improving the turbopump impeller efficiency, both are entirely based on open-source software
- The impeller shape has been converted in Bezier polynomials from data points. Twelve control points have been used as input variables for the optimization. The impeller efficiency η has been chosen as objective function and the head coefficient ψ has been used as constraint
- A single objective genetic algorithm optimization has been compared with a surrogate based optimization with ANN response surface

Conclusion and future work

- The surrogate based optimization method has shown almost the same optimum value provided by the standard optimization method, with saving the 37.5% of CFD simulations and therefore decreasing significantly the calculation time
- Although the high efficiency values predicted by the CFD simulations probably indicate the need of a more accurate CFD model, it seems reasonable to expect considerable room for efficiency improvement
- Different numerical methods for considering the rotor-stator interaction as well as new optimization strategies are planned for the next future, in order to further investigate the possibility of design improvement

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Thank You For Your Attention

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